

# Flutter-Enhanced Mixing: Flow-Induced Flutter of Flexible Membranes in Small Scale Mixers

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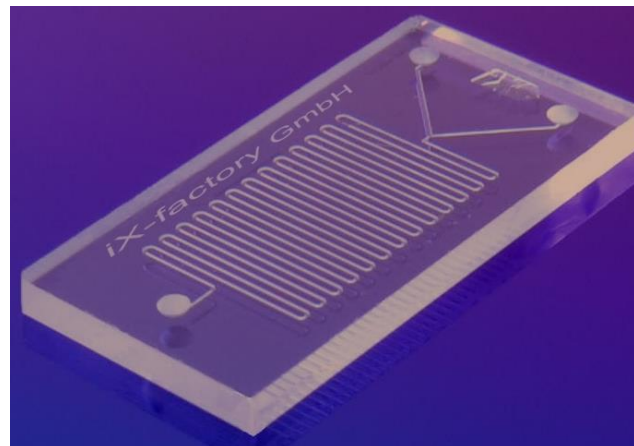
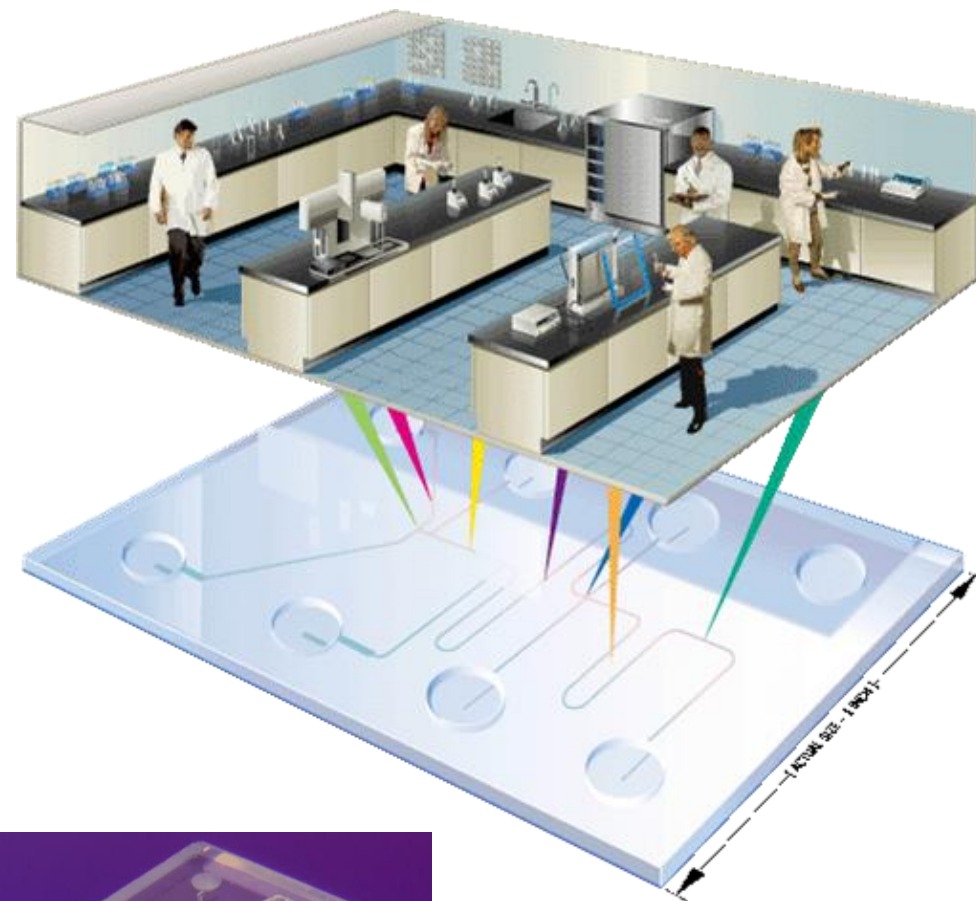
APS DFD 2019 Presentation



# Mixing Enhancement: Motivation



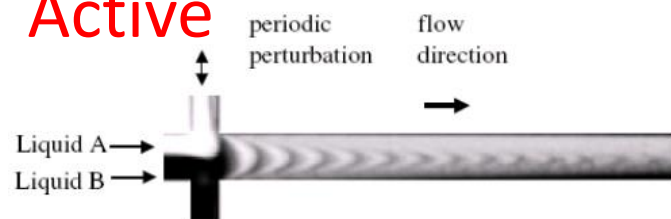
- Many low Re mixing applications such as lab-on-chip technology
- Microfluidic systems have many advantages over conventional approaches
- Inertial Microfluidics: ( $1 < \text{Re} < 100$ )
- Mixers usually classified as either active or passive mixers





# Mixing Enhancement: Active vs. Passive Mixing

## Active



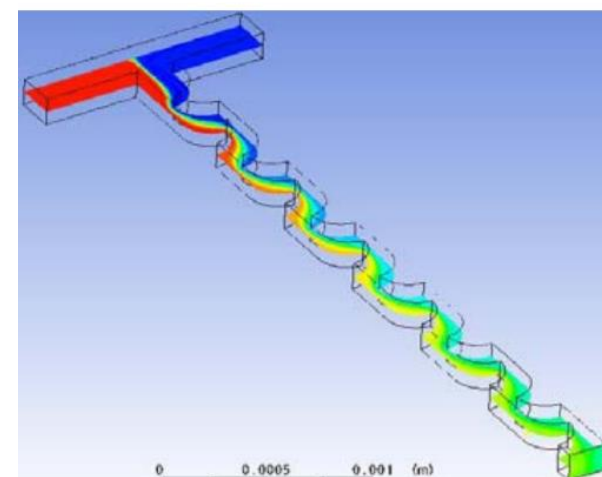
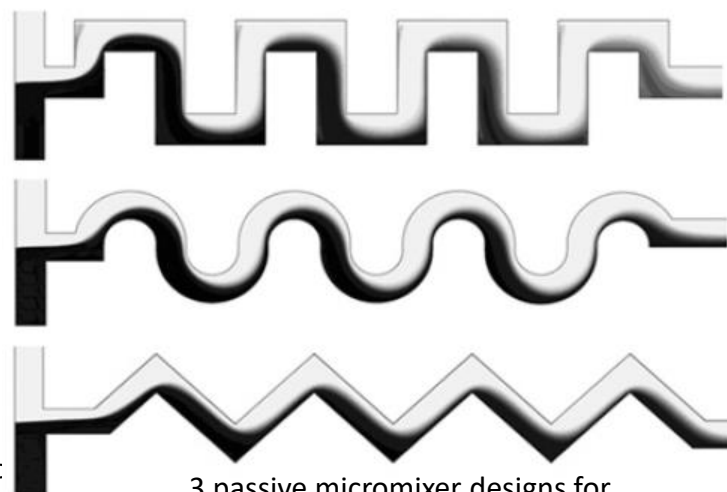
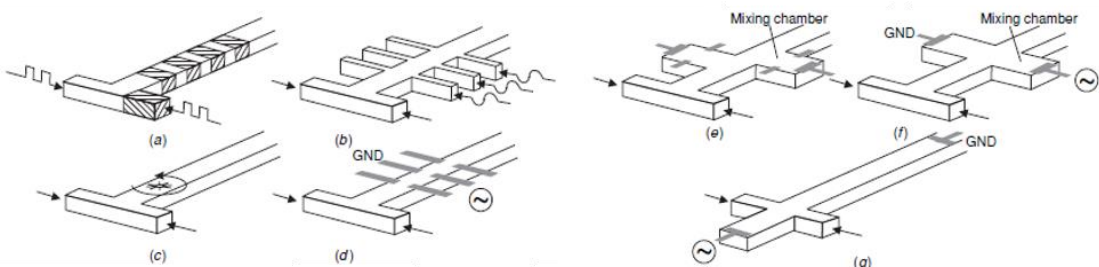
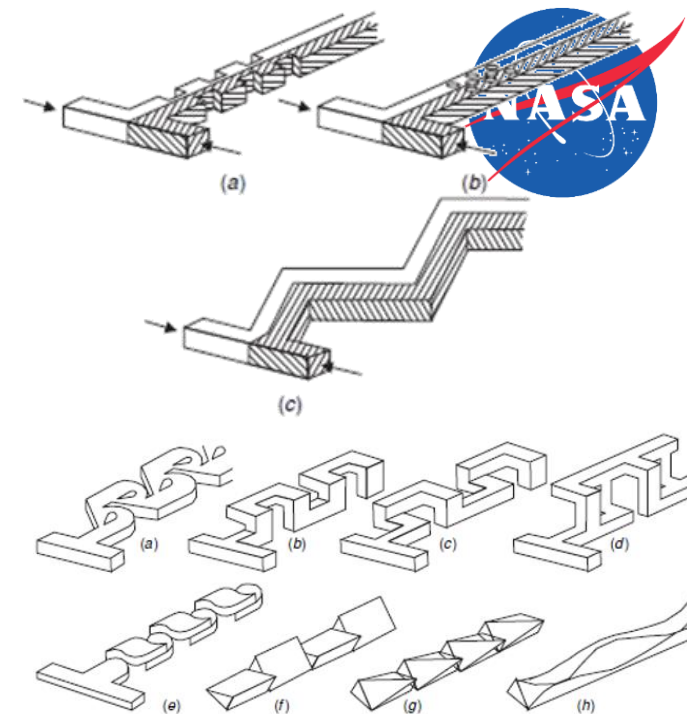
Periodic pressure perturbation  
design (Niu & Lee 2003)

- Many Approaches
- Good performance
- Complicated implementation

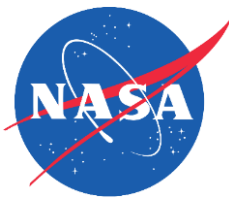
## Passive



- Inertial range allows for chaotic advection
- Typically poorer performance
- Simpler implementation





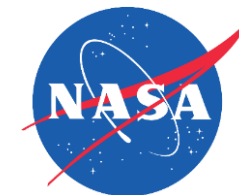


# Flow Induced Flutter (FIF)



- Flexible bodies immersed in a flow can undergo self excited aeroelastic flutter
- Flow induced vibration/flutter is ubiquitous in natural and engineered systems
- FIF is often a nuisance and even a failure mode
- However, we can take advantage of these phenomena





# Governing Equations

## Coupled Fluid – Structure – Scalar governing equations

Mass:

$$\nabla \cdot \mathbf{u} = 0$$

Momentum:

$$\frac{\partial \mathbf{u}}{\partial t} + \nabla \cdot (\mathbf{u}\mathbf{u}) = -\nabla p + \frac{1}{Re} \nabla^2 \mathbf{u} + \mathbf{f}$$

Incompressible Navier-Stokes  
on collocated\* cartesian grids

Position:

$$m_s \frac{\partial^2 \mathbf{X}}{\partial t^2} = \frac{\partial}{\partial s} (\sigma \boldsymbol{\tau} + \gamma \mathbf{n}) - \mathbf{F}$$

Zero thickness, elastic beam,  
inextensible (2D only)

Tension:

$$\frac{\partial^2}{\partial s^2} (\sigma \boldsymbol{\tau}) \cdot \boldsymbol{\tau} = \frac{m_s}{2} \frac{\partial^2}{\partial t^2} (\boldsymbol{\tau} \cdot \boldsymbol{\tau}) - m_s \frac{\partial \boldsymbol{\tau}}{\partial t} \cdot \frac{\partial \boldsymbol{\tau}}{\partial t} - \frac{\partial}{\partial s} \left( \frac{\partial (\gamma \mathbf{n})}{\partial s} - \mathbf{F} \right); \gamma = \frac{\partial}{\partial s} \left( k_b \frac{\partial^2 \mathbf{X}}{\partial s^2} \right) \cdot \mathbf{n}$$

Scalar:

$$\frac{\partial T}{\partial t} + \mathbf{u} \cdot \nabla T = \frac{1}{Re \cdot Sc} \nabla^2 T + q$$

Scalar Advection-Diffusion

Non-dimensional  
Parameters

$$Re = \frac{UL}{\nu}$$

$$U^* = UL \sqrt{\frac{m_s}{k_b}}$$

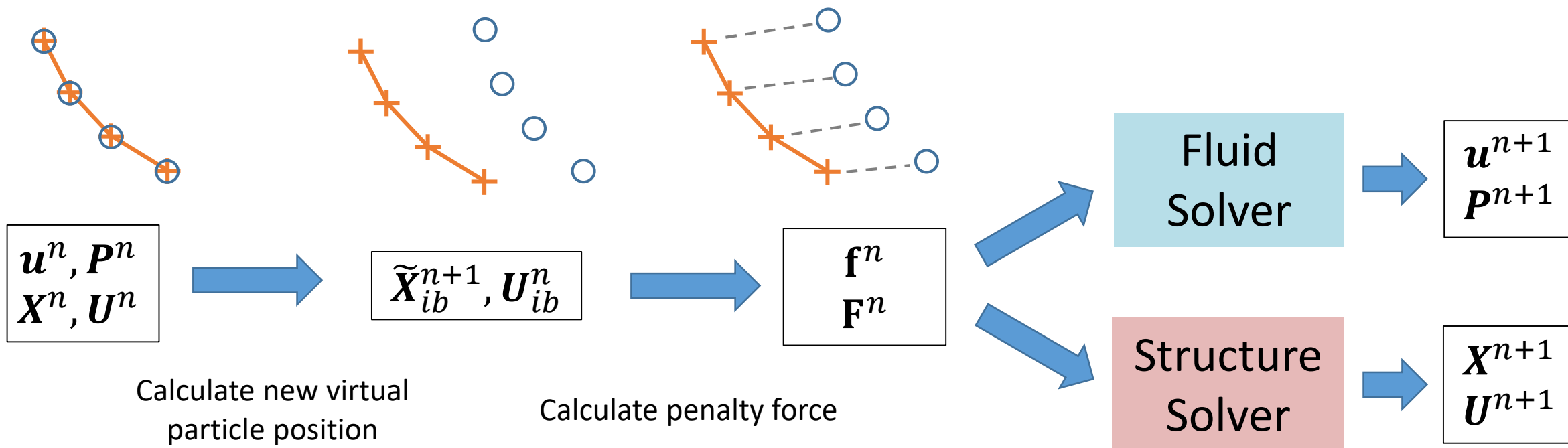
$$M^* = \frac{\rho L}{m_s}$$

$$Sc = \frac{\nu}{D}$$



# FSI Penalty Method Coupling

- + Lagrangian structure point
- Virtual fluid point
- Spring-damper "connector"



$$\tilde{X}_{ib}^{n+1} = X_{ib}^n + U_{ib}^n \Delta t$$

$$U_{ib}^n = \int_{\Omega_F} \mathbf{u}^n(\mathbf{x}) \delta[\mathbf{x} - \tilde{X}_{ib}^{n+1}] d\mathbf{x}$$

$$\mathbf{F}^n = -K_p \frac{\rho U^2}{L} [(\tilde{X}_{ib}^{n+1} - \mathbf{X}^n) + \beta \Delta t (\mathbf{U}_{ib}^n - \mathbf{U}^n)]$$

$$\mathbf{f}^n(\mathbf{x}) = \int_{\Gamma} \mathbf{F}^n(s) \delta[\mathbf{x} - \mathbf{X}^n(s)] ds$$

Note: Other bodies modeled using a sharp interface immersed boundary approach





# Mixing Enhancement: 2D Performance Measures

## • Interface Density ( $I_D$ )

- Calculate Normalized Gradient:
- $\nabla \phi_*(x, y) = \frac{\nabla \phi(x, y)}{\|\nabla \phi(x)\|_\infty}$
- Find peaks of  $\nabla \phi_*(x, y)$  using peak prominence to make  $\zeta$  field
- Average of  $\zeta$  at each  $x$  is  $I_D$



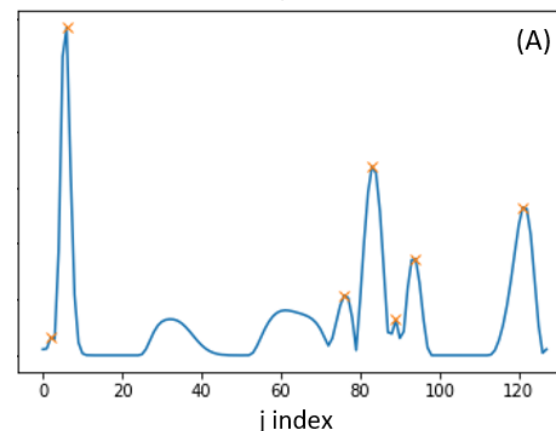
## • Head loss (HL)

- $H(x) = \int_0^H \left( P + \frac{1}{2} u^2 \right) dy$
- $HL(x) = H(0) - H(x)$
- Measures energy expended since the inlet

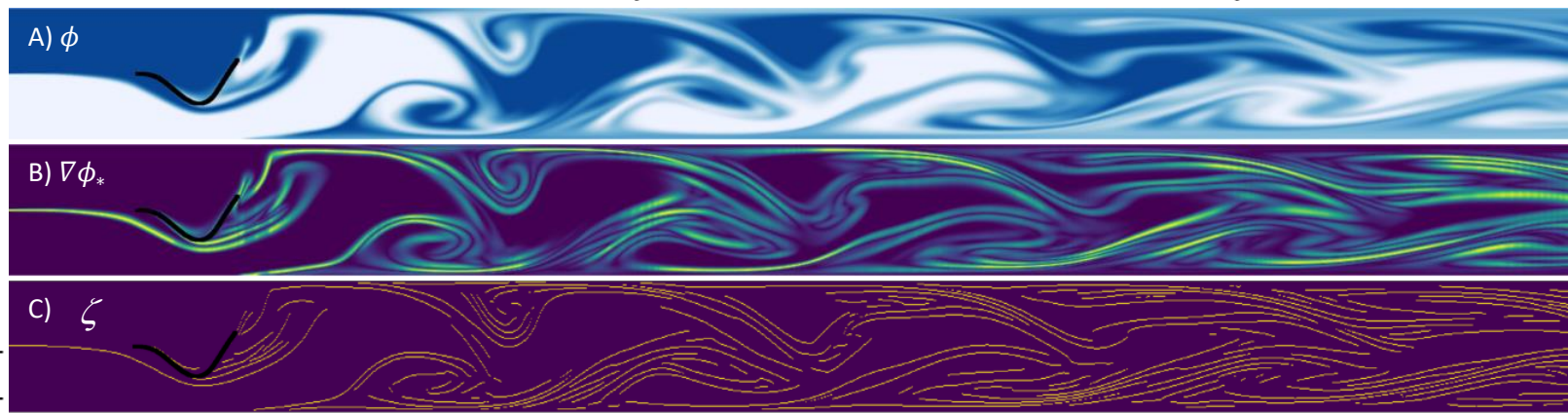
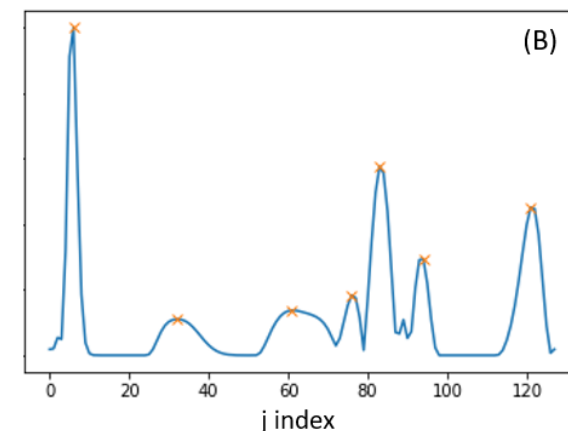
## • Mixing Index (M)

- $\sigma = \sqrt{\frac{1}{H} \int_0^H (T - \bar{T}_m)^2 dy}$
- $M = 1 - \sqrt{\frac{\sigma^2}{\sigma_m^2}}$
- M ranges from 0 to 1
  - 0 is unmixed flow ( $T = 0$  or 1)
  - 1 is mixed flow ( $T = 0.5$ )

Local Peak Amplitude Detection



Peak Prominence Detection



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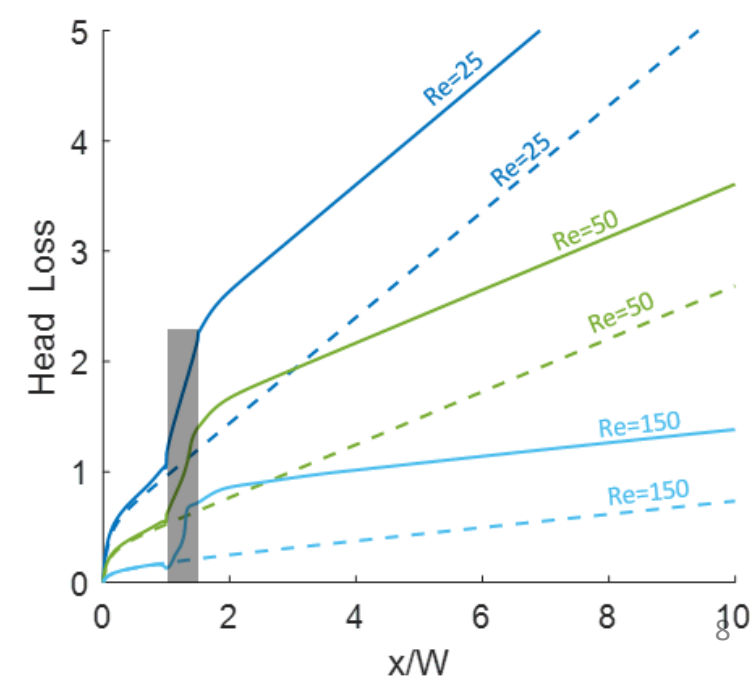
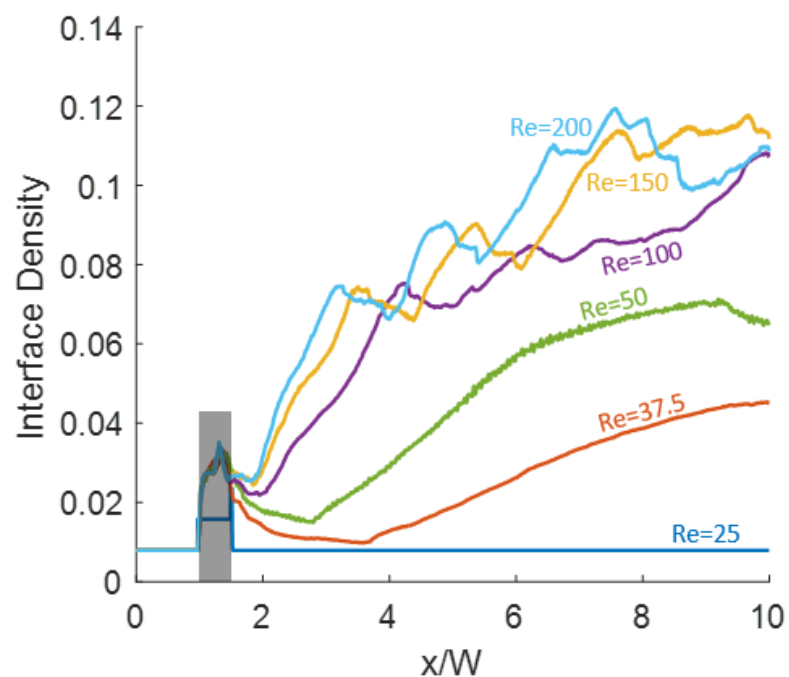
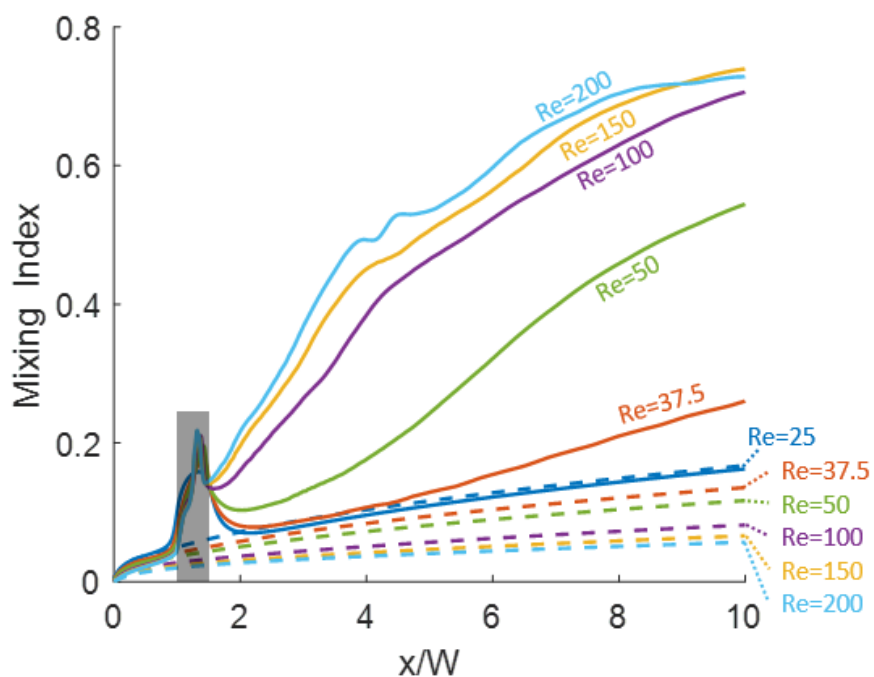
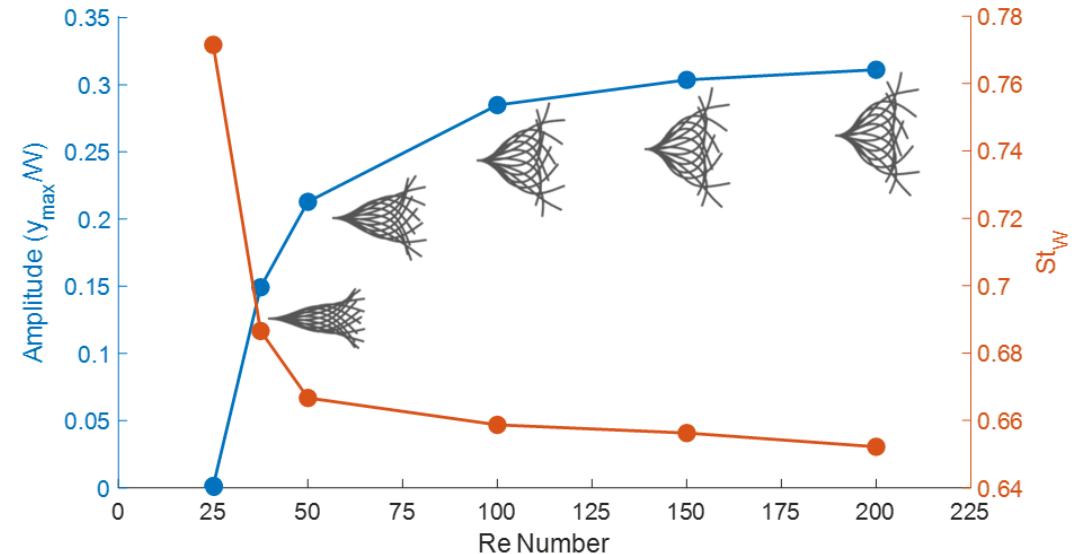
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# Effect of Reynolds Number

- Behavior progression: onset, transition, saturation
- Interface Density lags Mixing Index
- Nonmonotonicity in Interface Density is likely a 2D artifact
- One time Head Loss penalty due to flag



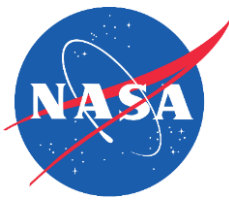




# Problem Description

- $Re = 50-200$ ;  $Sc = 100$
- $dx_f = 0.015L$ ;  $960 \times 96 \times 96$
- $dx_s = 0.0125L$ ; 120 pts/ $L$
- $dt = 0.0001$ ;  $10^6$  time steps

$$\frac{\partial \mathbf{u}}{\partial t} + \nabla \cdot (\mathbf{u}\mathbf{u}) = -\nabla p + \frac{1}{Re} \nabla^2 \mathbf{u} + \mathbf{f}$$

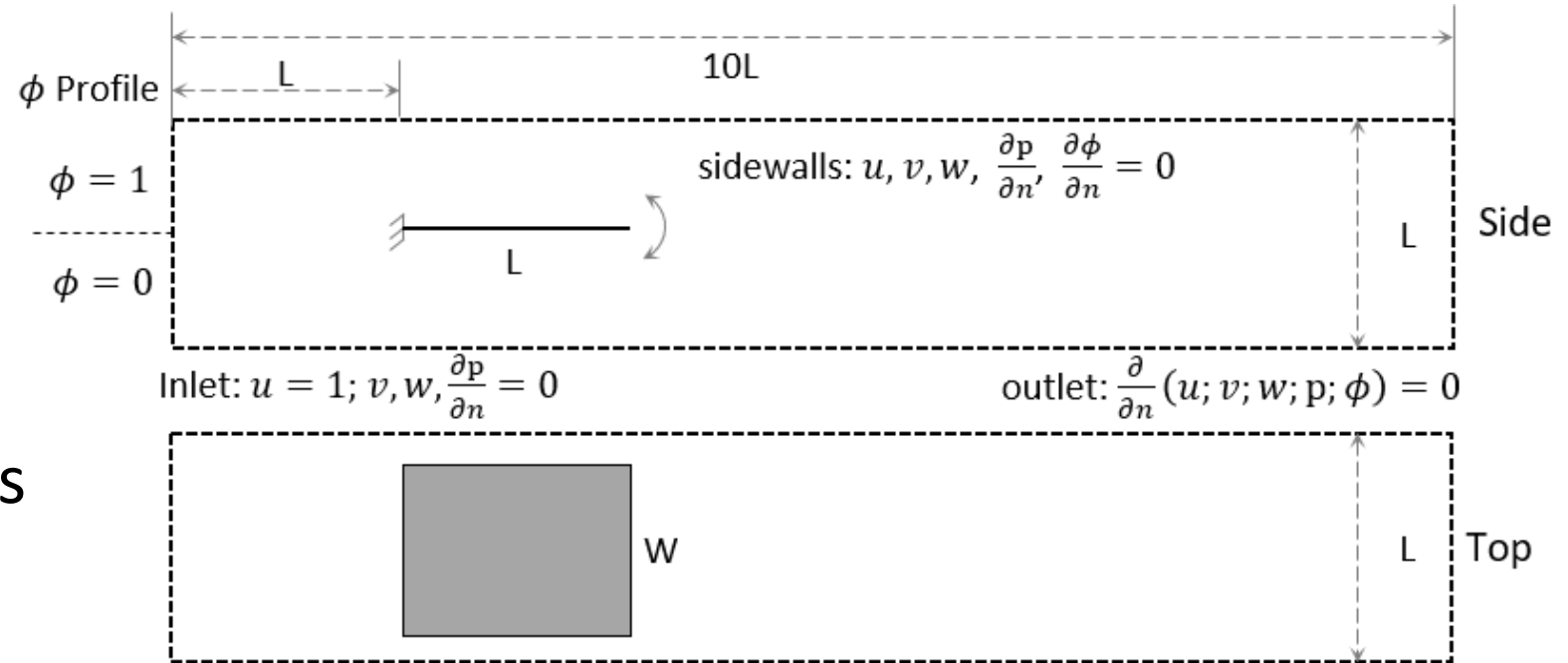


Governing  
Equations:

$$\nabla \cdot \mathbf{u} = 0$$

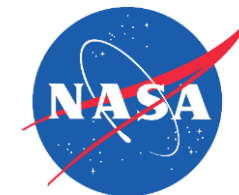
$$\frac{\partial T}{\partial t} + \mathbf{u} \cdot \nabla T = \frac{1}{Re \cdot Pr} \nabla^2 T$$

$$\frac{\partial^2 \mathbf{X}}{\partial t^2} = \sum_{i,j=1}^2 \left[ \frac{\partial}{\partial s_i} \left( \sigma_{ij} \frac{\partial \mathbf{X}}{\partial s_j} \right) - \frac{\partial^2}{\partial s_i \partial s_j} \left( \frac{M^*}{U^2} \frac{\partial^2 \mathbf{X}}{\partial s_i \partial s_j} \right) \right] + Fr \frac{\mathbf{g}}{g} - \mathbf{F}$$





# Performance Measures



## Head loss (HL)

$$H^*(x) = \frac{1}{\rho U^2} \int_0^Z \int_0^Y \left( P + \frac{1}{2} u^2 \right) dy dz$$

$$HL(x) = H^*(0) - H^*(x)$$

## Interface Density ( $I_D$ )

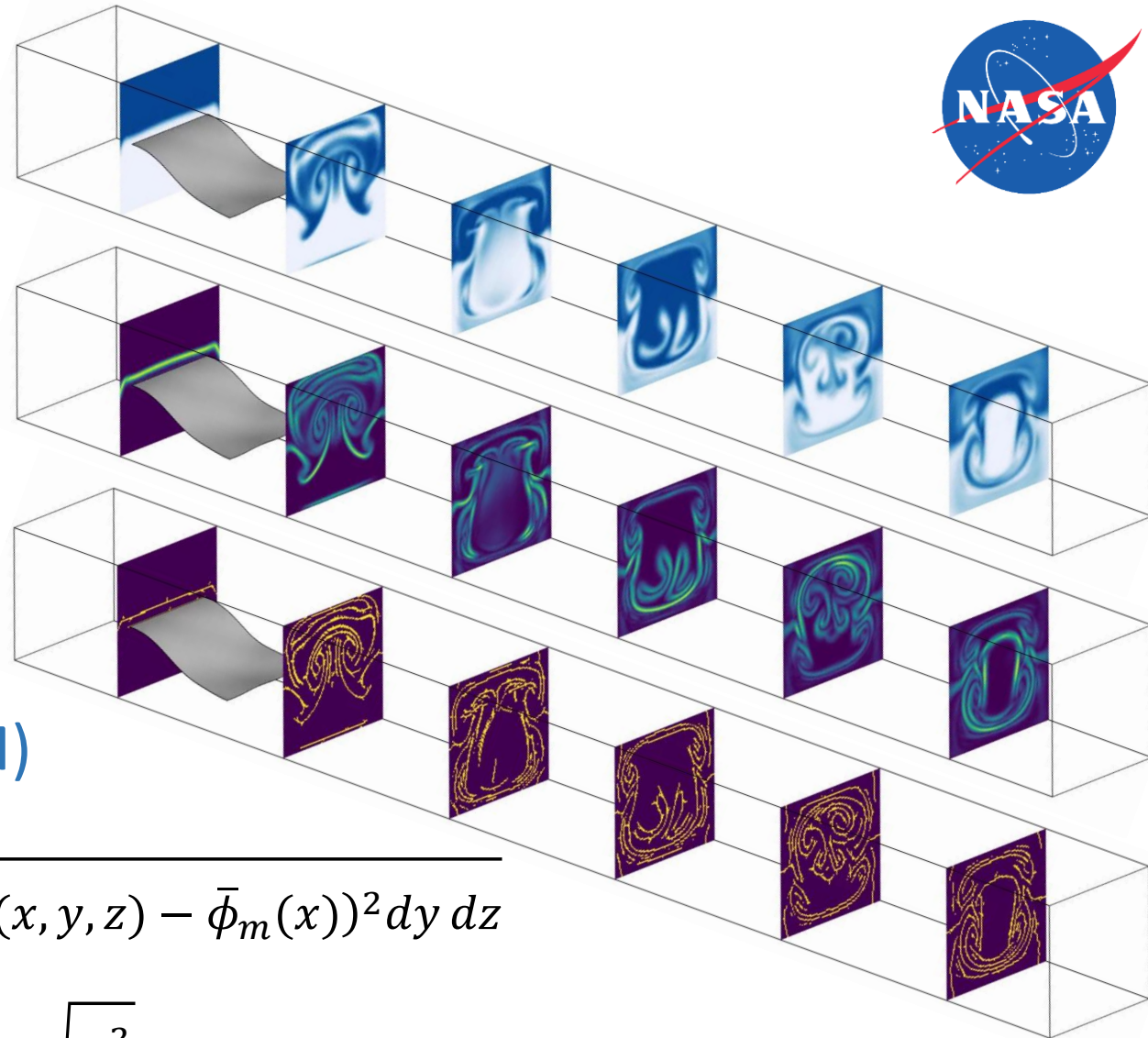
$$\nabla \phi_*(x, y, z) = \frac{\nabla \phi(x, y, z)}{\|\nabla \phi(x)\|_\infty}$$

$$I_D(x) = \frac{1}{L^2} \sum_{z=0}^Z \sum_{y=0}^Y \zeta(x, y, z)$$

## Mixing Index (M)

$$\sigma = \sqrt{\frac{1}{L^2} \int_0^Z \int_0^Y (\phi(x, y, z) - \bar{\phi}_m(x))^2 dy dz}$$

$$M = 1 - \sqrt{\frac{\sigma^2}{\sigma_m^2}}$$



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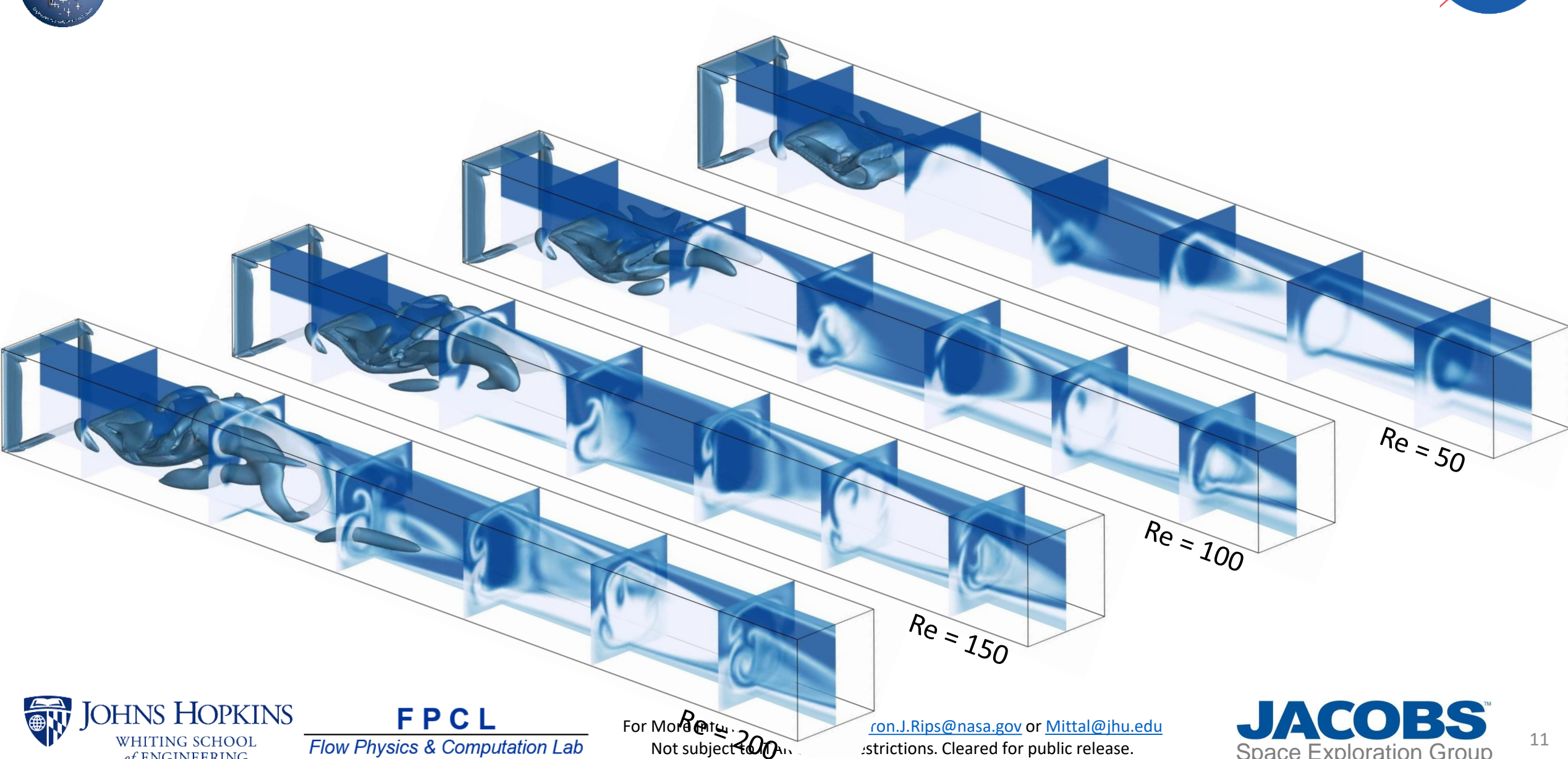
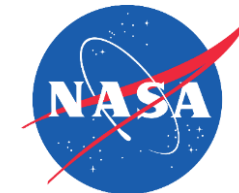
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# 3D Visualizations



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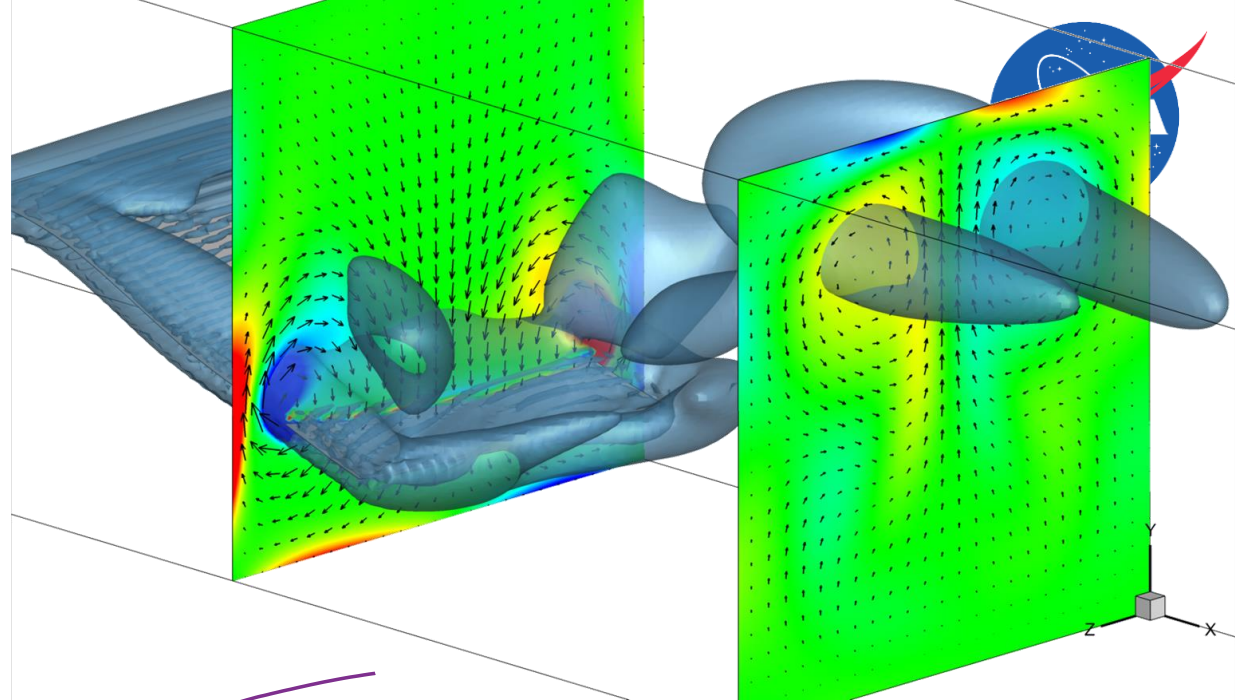
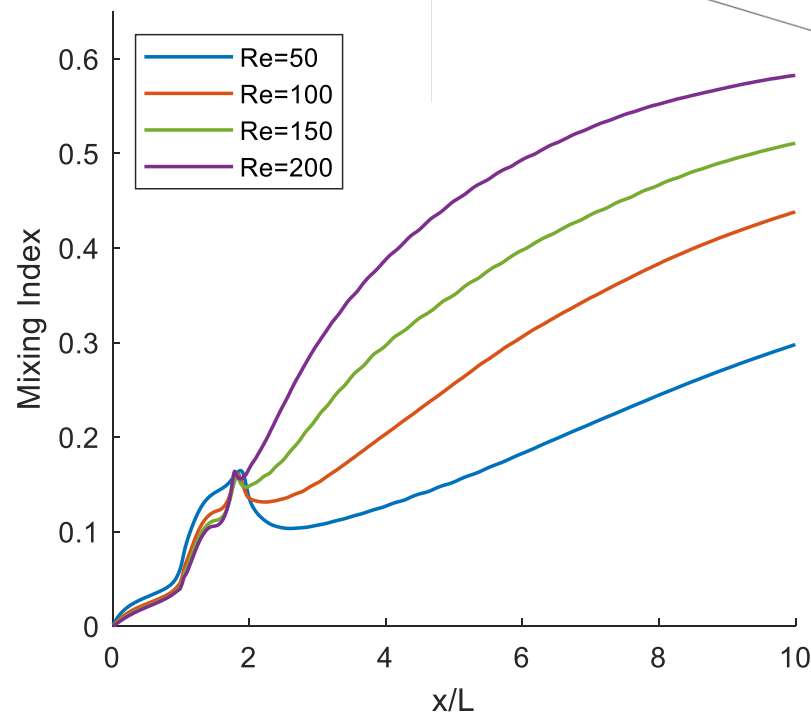
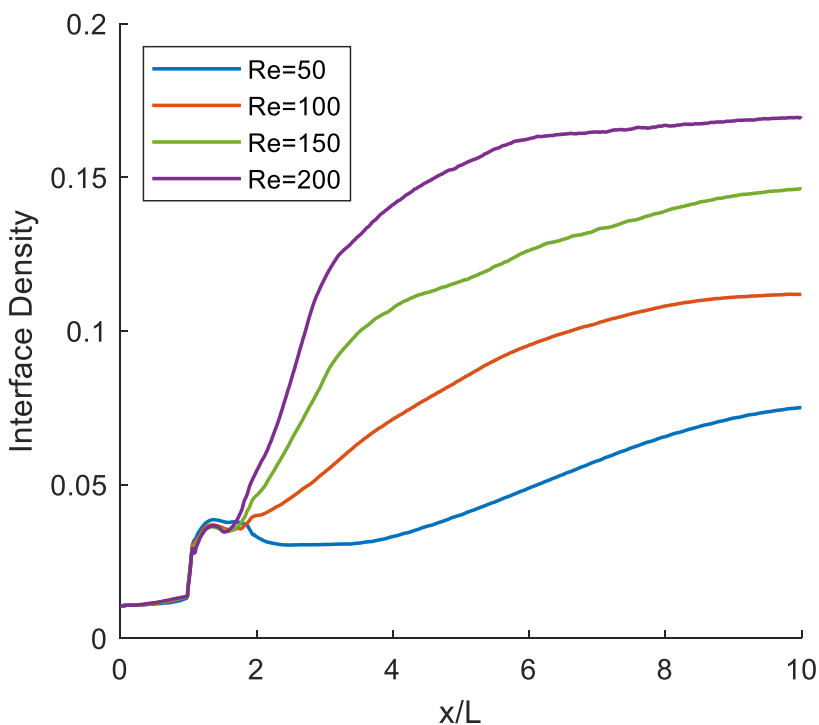
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# Mixing Mechanism and Performance

Interface Density increase lead Mixing Index improvements



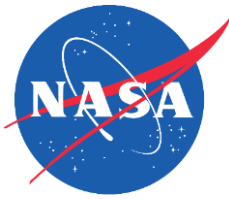
Streamwise vortices are the leading edge vortices on the streamwise edges of the flag

These horseshoe vortices are typical of 3D flapping flags

The confinement aligns the horseshoe in the flow direction, making it more persistent







# Acknowledgements

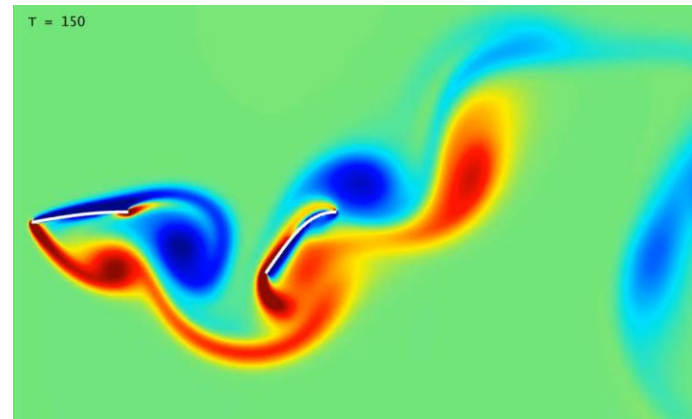
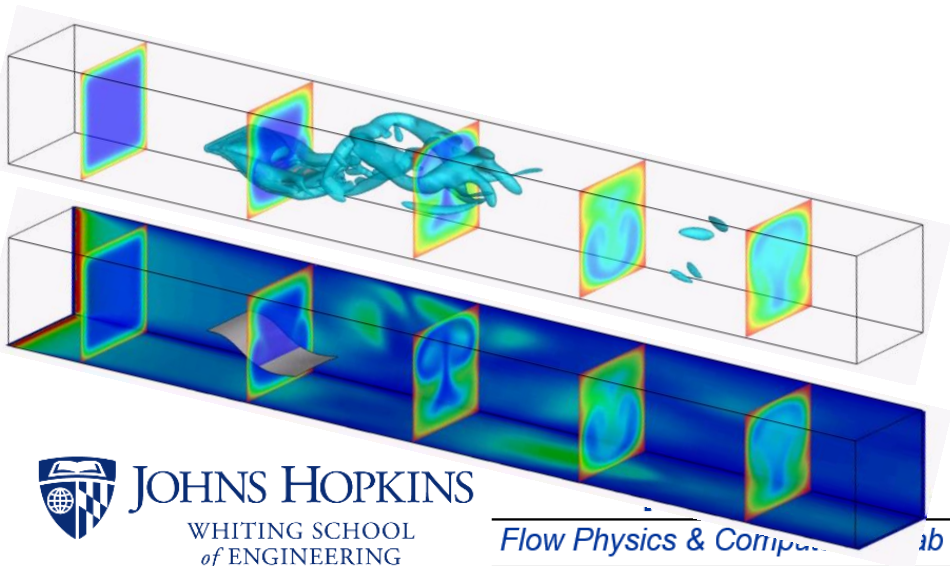
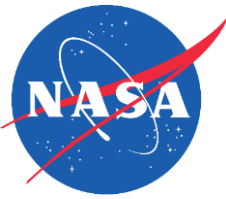
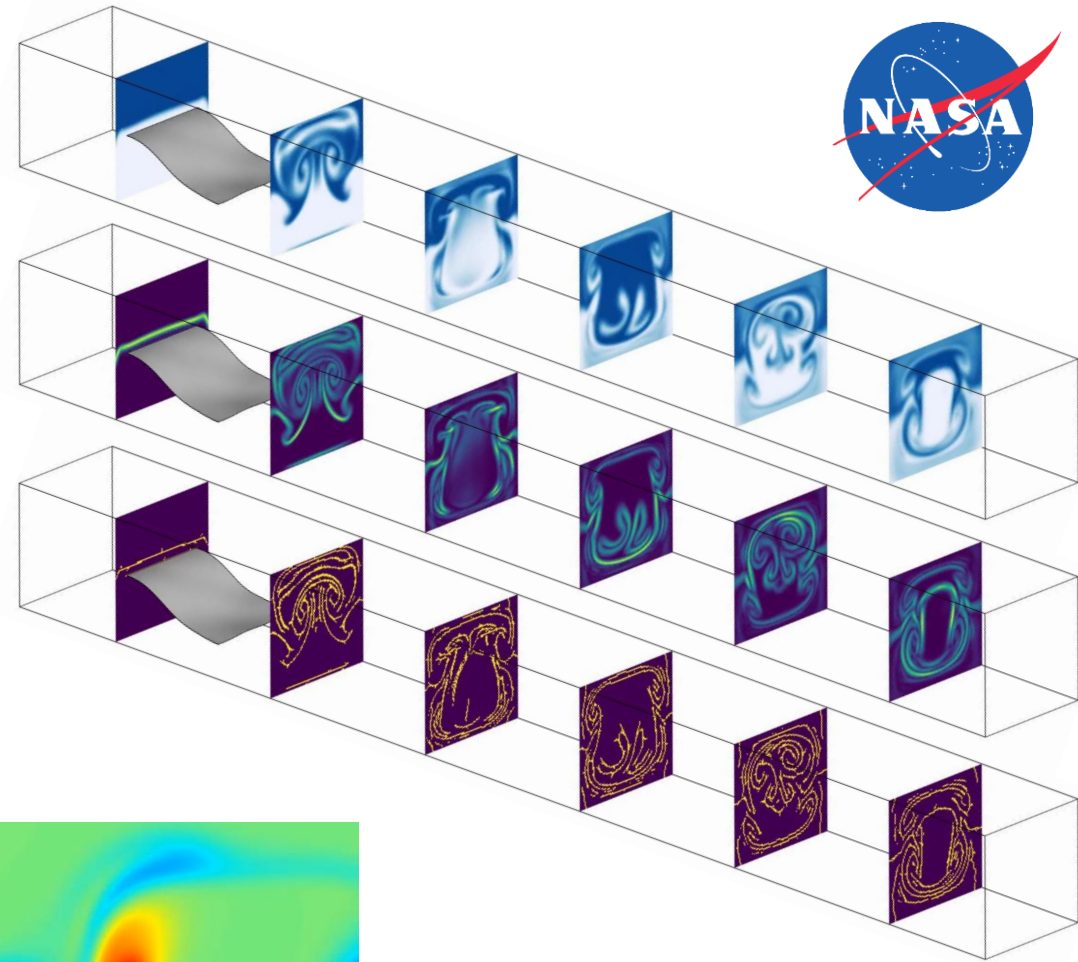
- Sponsors: NSF, EPRI, AFOSR
- JHU engineering community
- Lab mates
- Professors, teachers, mentors
- My advisor Prof. Rajat Mittal





# Conclusions

- Flow Induced Flutter can be harnessed for engineering applications
  - Heat Transfer, Mixing, and Energy Harvesting
- Taking advantage of a Passive-Unsteady flow control strategy



## Questions?

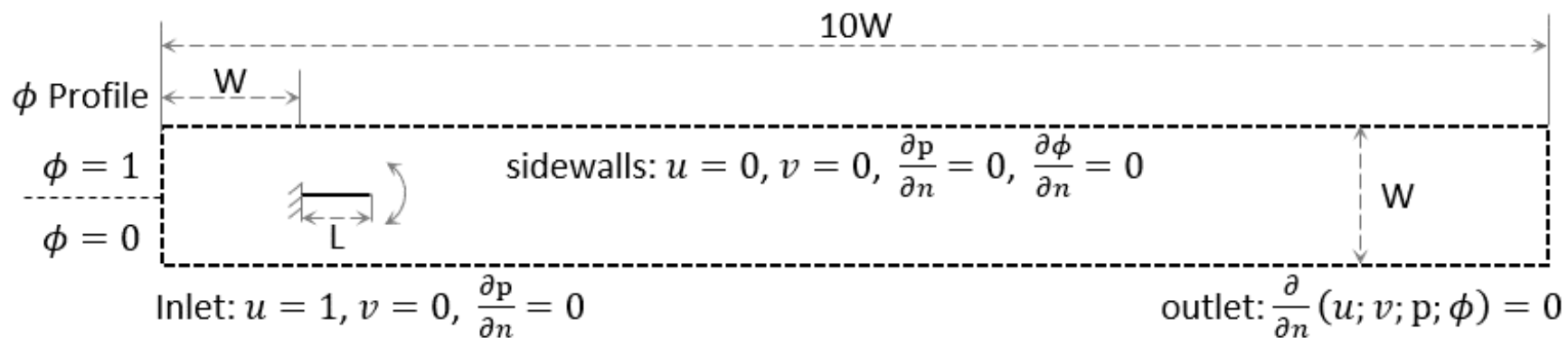
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# Backup Slides



# Mixing Enhancement: 2D Problem Description



- Uniform fluid grid:
  - $dx = dy = 0.016L$
  - 1280x128 points
- Structure grid:
  - $ds = 0.012L$  or 80 points
- Temporal resolution:
  - $dt = 1.0 \times 10^{-4}$
  - $10^6$  time steps

$$\frac{\partial \mathbf{u}}{\partial t} + \nabla \cdot (\mathbf{u}\mathbf{u}) = -\nabla p + \frac{1}{\text{Re}} \nabla^2 \mathbf{u} + \mathbf{f}$$

$$\nabla \cdot \mathbf{u} = 0$$

$$\frac{\partial \phi}{\partial t} + \mathbf{u} \cdot \nabla \phi = \frac{1}{\text{Re} \cdot \text{Sc}} \nabla^2 \phi$$

$$\frac{\partial^2 \mathbf{X}}{\partial t^2} = \frac{\partial}{\partial s} \left( \varsigma \frac{\partial \mathbf{X}}{\partial s} \right) - \frac{\partial^2}{\partial s^2} \left( \frac{M^*}{U^{*2}} \frac{\partial \mathbf{X}}{\partial s} \right) - M^* \mathbf{F}$$

$$\frac{\partial \mathbf{X}}{\partial s} \cdot \frac{\partial^2}{\partial s^2} \left( \varsigma \frac{\partial \mathbf{X}}{\partial s} \right) = \frac{1}{2} \frac{\partial^2}{\partial t^2} \left( \frac{\partial \mathbf{X}}{\partial s} \cdot \frac{\partial \mathbf{X}}{\partial s} \right) - \frac{\partial^2 \mathbf{X}}{\partial t \partial s} \cdot \frac{\partial^2 \mathbf{X}}{\partial t \partial s} - \frac{\partial \mathbf{X}}{\partial s} \cdot \frac{\partial}{\partial s} (\mathbf{F})$$

## Parametric Sweep

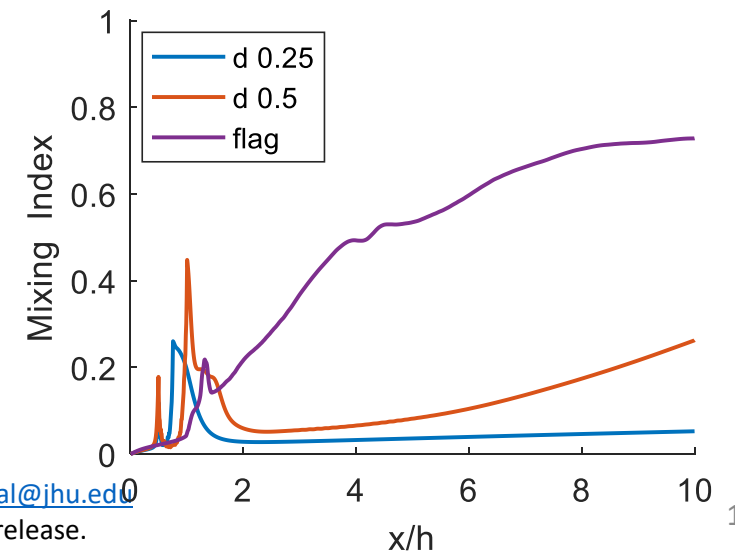
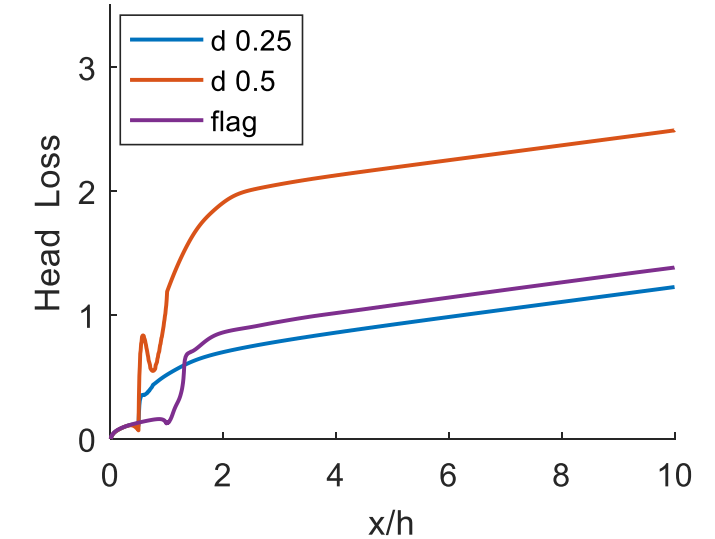
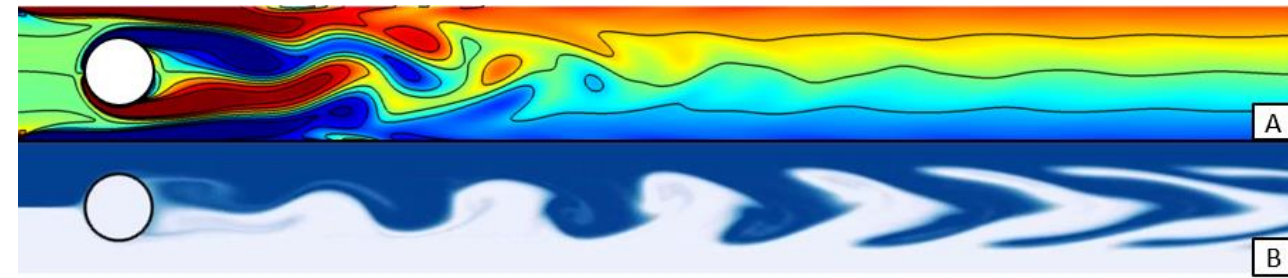
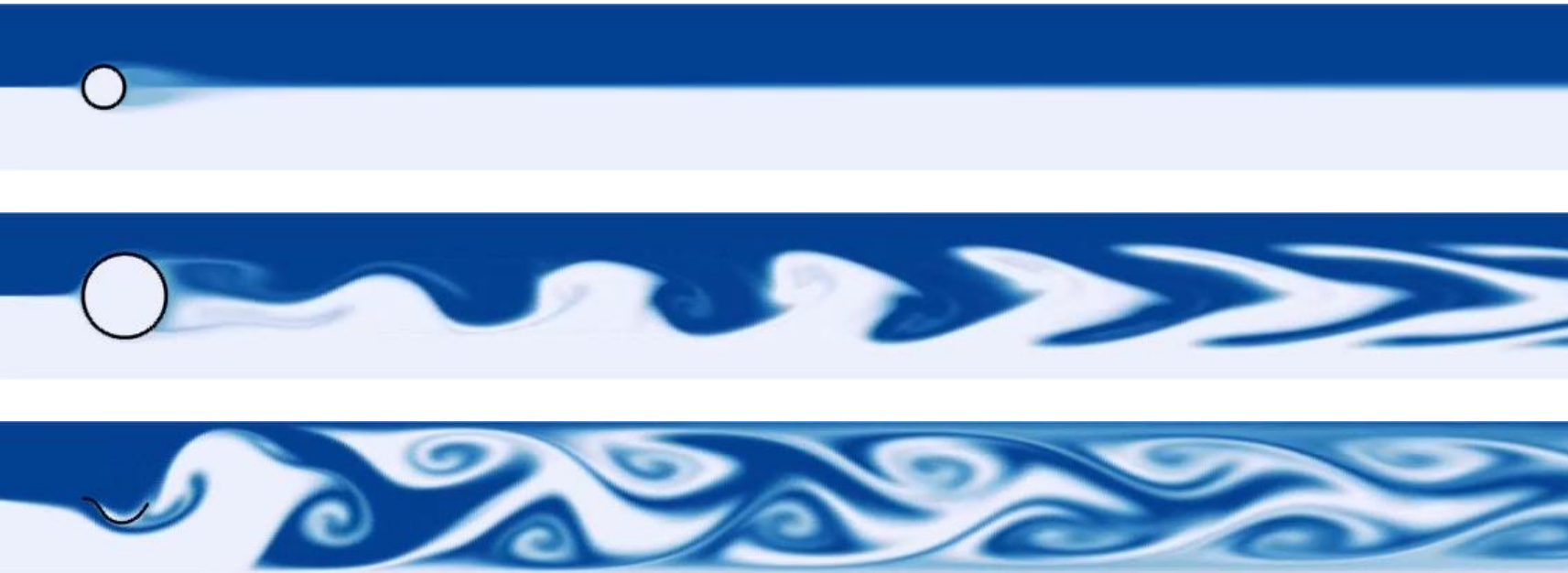
- $L=W/2$  Re: 25, 37.5, 50, 100, 150, 200
- $L=W$  Re: 15, 20, 25, 50, 100, 150, 200
- $L=2W$  Re: 10, 15, 20, 25
- Sc sweep ( $L=W/2$ , Re=200): 1, 10, 100, 1000





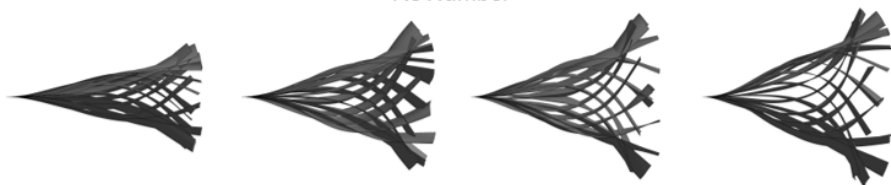
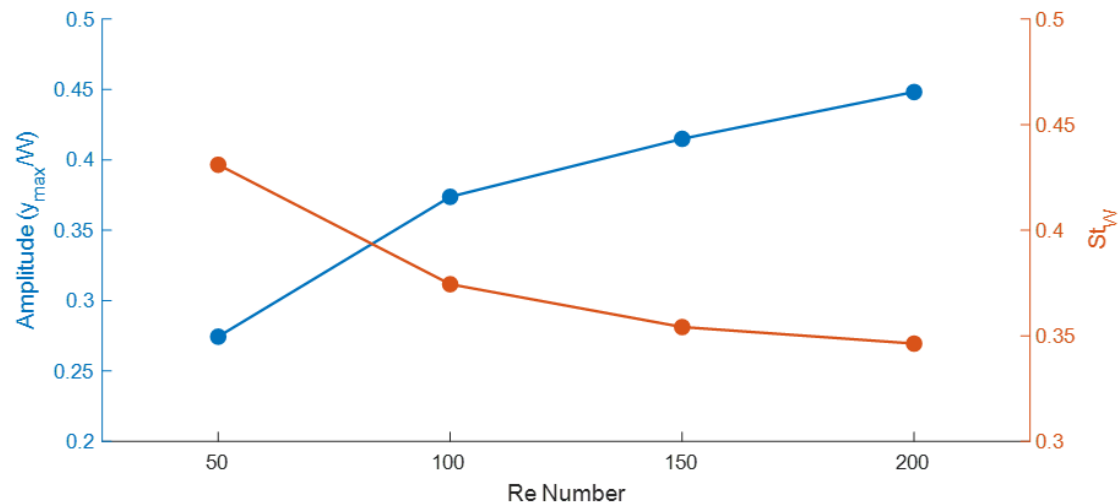
# Mixing Enhancement: 2D Bluff Body Mixing

A single flapping membrane is meant as a single unit of a larger system. In which case it is analogous to a single cylinder in a post mixer





# Effect of Reynolds Number

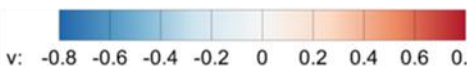
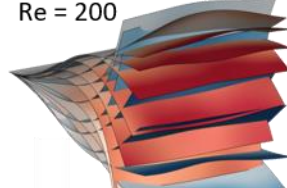
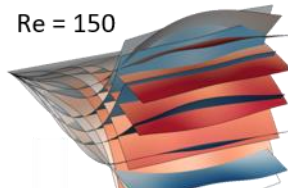
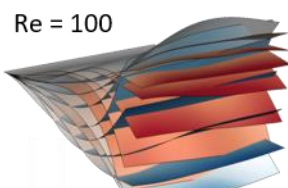
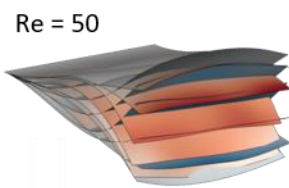


Re = 50

Re = 100

Re = 150

Re = 200



v: -0.8 -0.6 -0.4 -0.2 0 0.2 0.4 0.6 0.8

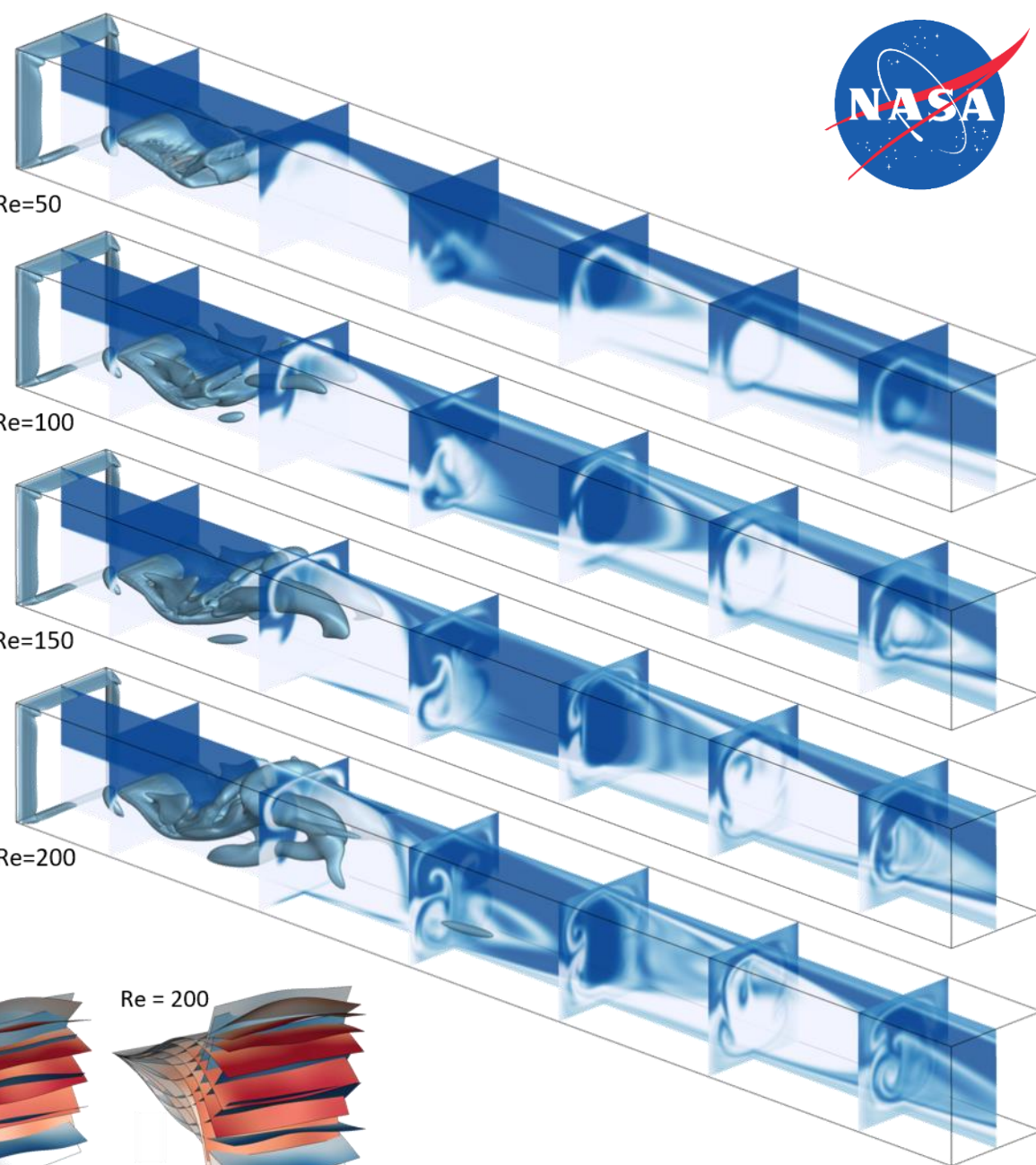
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A) Re=50

B) Re=100

C) Re=150

D) Re=200



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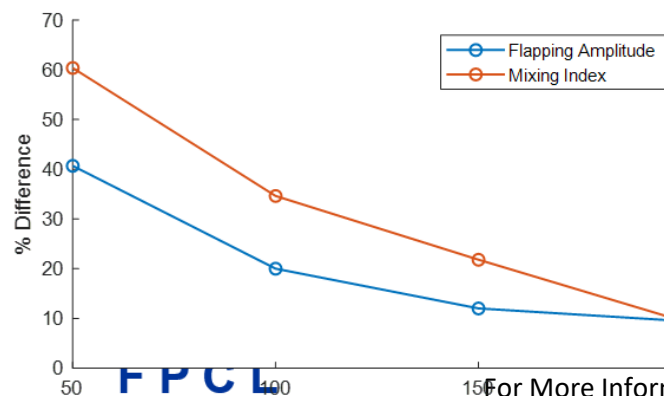
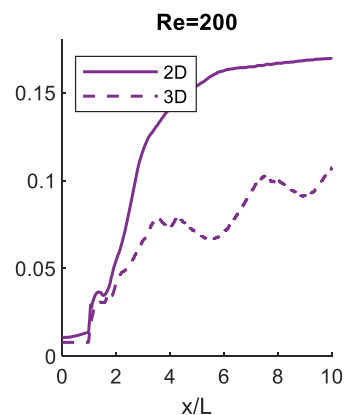
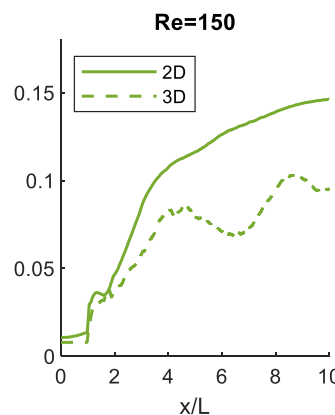
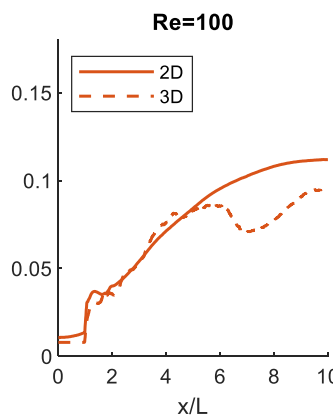
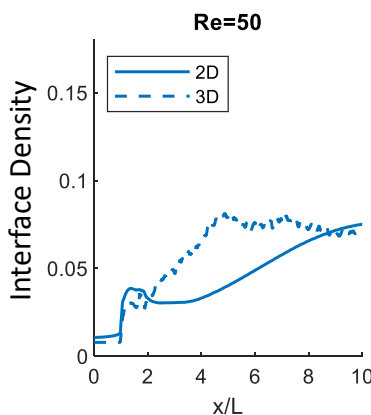
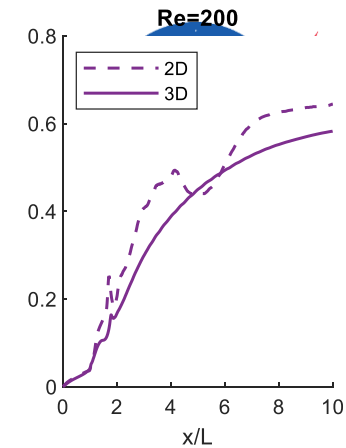
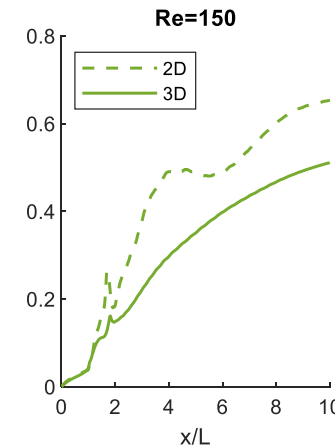
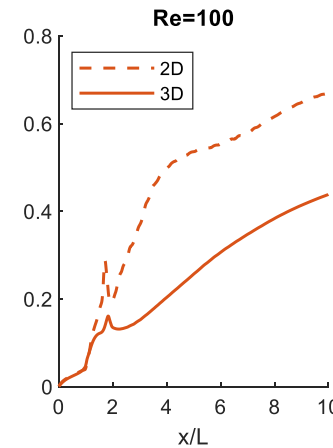
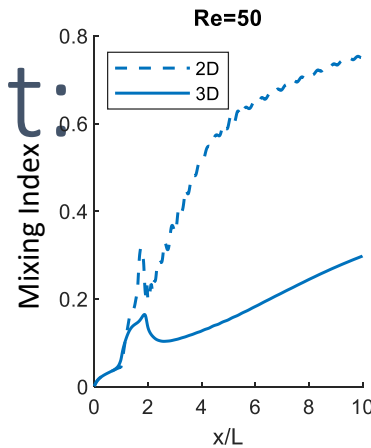
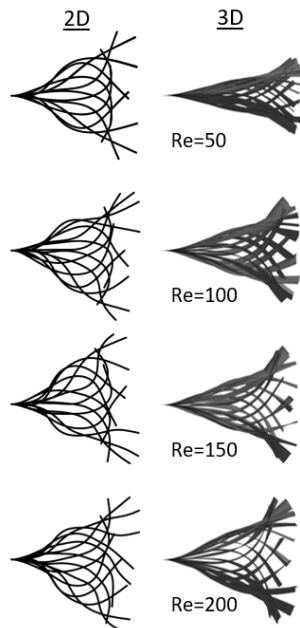
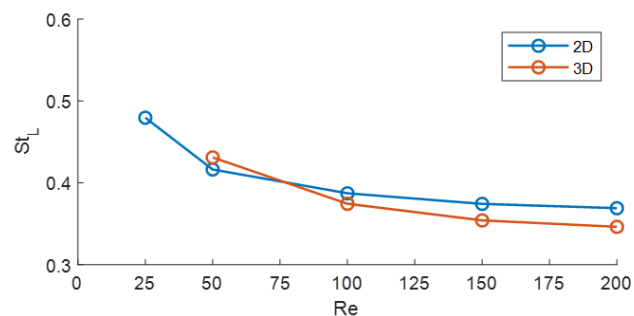
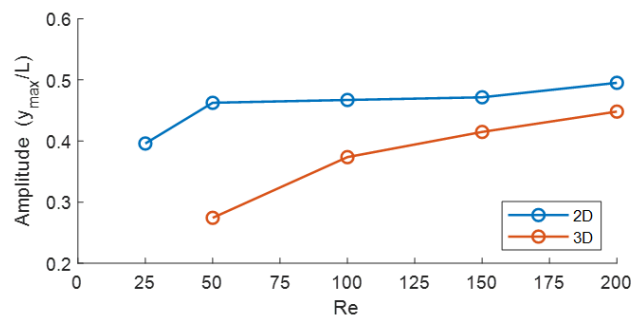
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# Mixing Enhancement: 2D vs. 3D



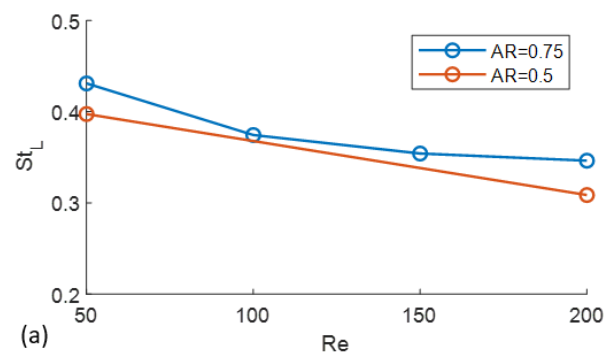
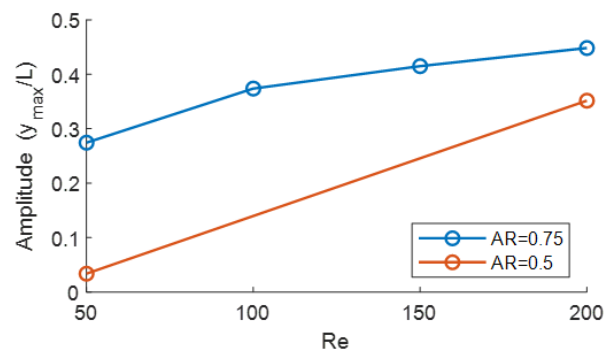
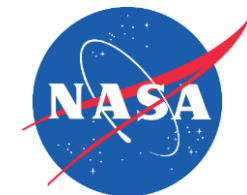
Marginal Correlation  
between amplitude  
difference and mixing  
index difference



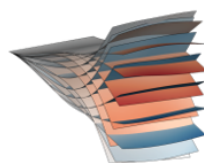
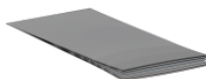




# Confinement and Spanwise Gap

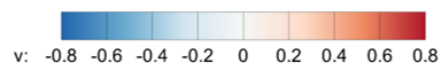
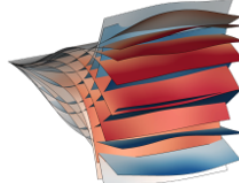
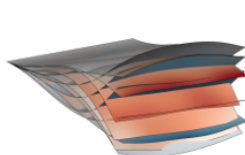


Re = 50, AR=0.5      Re = 200, AR=0.5



Re = 50, AR=0.75

Re = 200, AR=0.75



Re

50

AR=0.5

AR=0.75

200

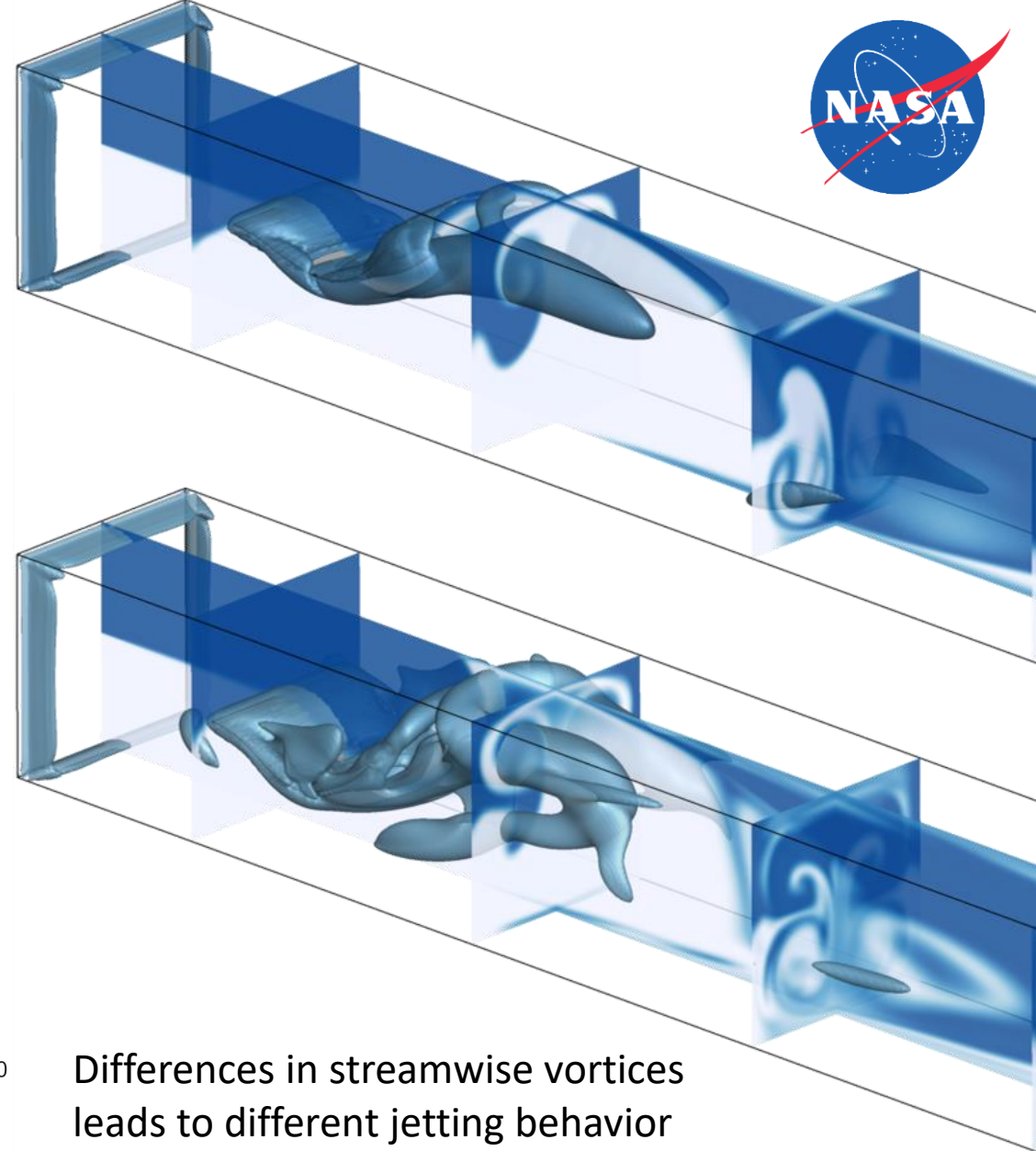
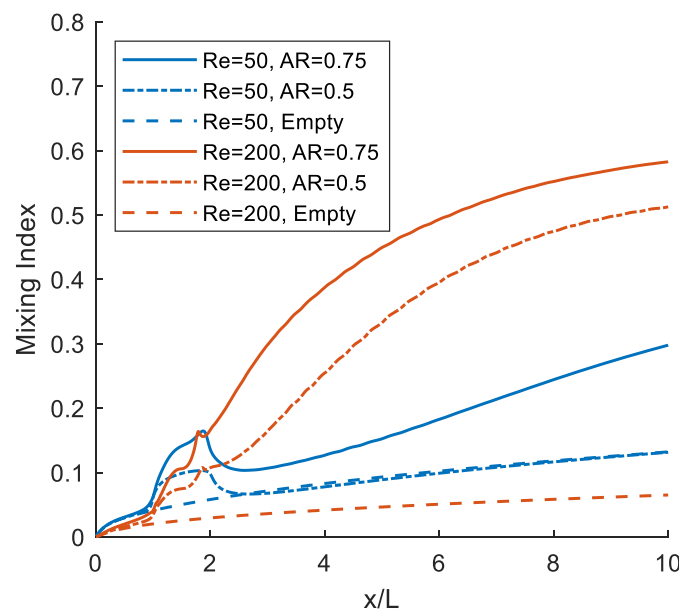
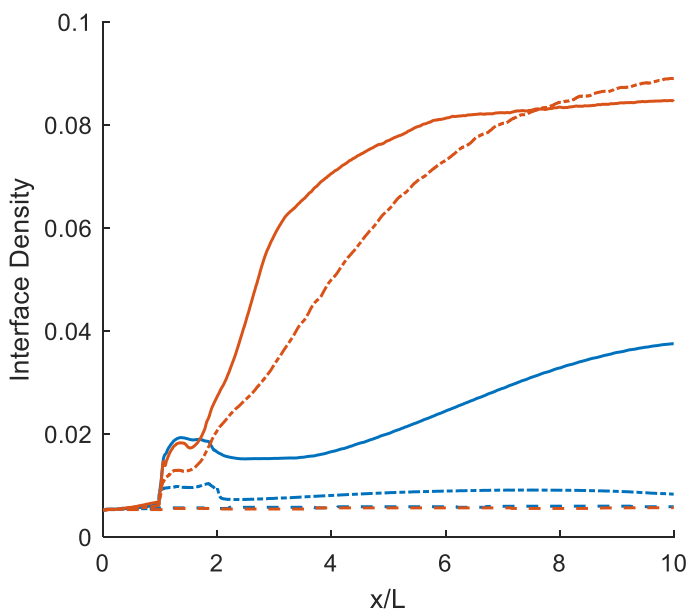
AR=0.5

AR=0.75





# Confinement and Spanwise Gap



Differences in streamwise vortices leads to different jetting behavior